

# Waves and Compressible Flow

## Lecture 3

## 2. Models for linear wave propagation

p.1

- Various examples:
  - acoustic waves in a gas
  - gravity waves on an interface.
  - internal gravity waves in a stratified fluid
  - inertial waves in a rotating fluid
  - elastic waves
- Key assumption is that waves are small amplitude, which allows linearization of the governing equations.

# Acoustic waves

P.2

- Consider barotropic flow with no body force:

$$\frac{D\rho}{Dt} = -\rho \nabla \cdot \underline{u} \quad (1)$$

$$\rho \frac{D\underline{u}}{Dt} = -\nabla p \quad (2)$$

$$p = P(\rho) \quad (3)$$

- Stationary state:  $\underline{u} = \underline{0}$ ,  $\rho = \rho_0$ ,  $p = p_0 = P(\rho_0)$

- Introduce small perturbation:  $\underline{u} = \underline{0} + \underline{u}'$ ,  $\rho = \rho_0 + \rho'$ ,  $p = p_0 + p'$

- Substitute into ①-③ and linearize about stationary state by neglecting quadratic and higher order terms.

$$\textcircled{1} \Rightarrow \frac{\partial \rho'}{\partial t} + (\underline{u}' \cdot \nabla) \rho' = -(\rho_0 + \rho') \nabla \cdot \underline{u}' \xrightarrow{\text{(linearizing)}} \frac{\partial \rho'}{\partial t} = -\rho_0 \nabla \cdot \underline{u}' \quad \textcircled{1L}$$

$$\textcircled{2} \Rightarrow (\rho_0 + \rho') \left( \frac{\partial \underline{u}'}{\partial t} + (\underline{u}' \cdot \nabla) \underline{u}' \right) = -\nabla p' \xrightarrow{\text{(linearizing)}} \rho_0 \frac{\partial \underline{u}'}{\partial t} = -\nabla p' \quad \textcircled{2L}$$

$$\textcircled{3} \Rightarrow \rho_0 + \rho' = P(\rho_0 + \rho') = P(\rho_0) + \frac{dP}{d\rho}(\rho_0) \rho' + \dots \xrightarrow{\text{(linearizing)}} \rho' = c_0^2 \rho' \quad \textcircled{3L}$$

(Taylor expanding)

where  $c_0^2 = \frac{dP}{d\rho}(\rho_0)$ , e.g. isentropic flow  $p = p(\rho) \Rightarrow c_0^2 = \frac{\gamma p_0}{\rho_0}$ .

•  $\nabla \cdot \underline{u}' = 0 \Rightarrow \frac{\partial}{\partial t} (\nabla \cdot \underline{u}') = 0$ , so if we assume  $\nabla \cdot \underline{u}' = 0$  initially, then  $\nabla \cdot \underline{u}' = 0$  for  $t \geq 0$  and  $\exists \phi(\underline{x}, t)$  s.t.  $\underline{u}' = \nabla \phi$  (4L)

• Then (2L)  $\Rightarrow \nabla \cdot (\rho_0 \frac{\partial \phi}{\partial t} + p') = 0 \Rightarrow \rho_0 \frac{\partial \phi}{\partial t} + p' = F(t) = 0$  (5L)  
wlog, which is a linearized Bernoulli equation.

• Hence,  $\rho_0 \phi_{tt} \stackrel{(5L)}{=} -p'_t$ ,  $p'_t \stackrel{(3L)}{=} c_0^2 p'_t$  and  $p'_t \stackrel{(1L)}{=} -\rho_0 \nabla \cdot \underline{u}' \stackrel{(5L)}{=} -\rho_0 \nabla^2 \phi$

$$\Rightarrow \phi_{tt} = c_0^2 \nabla^2 \phi \quad (6L)$$

(Wave equation, wave speed  $c_0$ )

• (5L)  $\Rightarrow p' = -\rho_0 \phi_t$ , so  $\frac{\partial}{\partial t}$  (6L)  $\Rightarrow p'_{tt} = c_0^2 \nabla^2 p'$

• (3L)  $\Rightarrow p' = c_0^2 \rho'$ , so  $p'_{tt} = c_0^2 \nabla^2 \rho'$

• (4L)  $\Rightarrow \underline{u}' = \nabla \phi$ , so  $\nabla$  (6L)  $\Rightarrow \underline{u}'_{tt} = c_0^2 \nabla^2 \underline{u}'$

• After linearization everything is governed by the same wave equation.

• NB: For air  $p = c \rho^\gamma$  with  $\gamma \approx 1.4 \Rightarrow c_0 \approx 340 \text{ m s}^{-1}$ , which agrees well with measurements.

## Example: 1D waves

- $p' = p'(x, t) \Rightarrow p'_{tt} = c_0^2 p'_{xx} \Rightarrow p' = \underbrace{F(x - c_0 t)}_{\text{right travelling wave}} + \underbrace{G(x + c_0 t)}_{\text{left travelling wave}}$

- Consider a transducer at  $x=0$  imposing a periodic pressure fluctuation s.t.

$$p'(0, t) = a \cos \omega t + b \sin \omega t = \operatorname{Re}(A e^{-i\omega t})$$

where  $\omega$  is frequency and  $A = a + ib$  is complex amplitude.

- Try separable time-harmonic ansatz  $p' = f(x) e^{-i\omega t}$  (real part understood), where the complex-valued function  $f(x)$  is TBD.

• Wave equation  $\Rightarrow \frac{d^2 f}{dx^2} + \frac{\omega^2}{c_0^2} f = 0 \Rightarrow f = \alpha e^{i\omega x/c_0} + \beta e^{-i\omega x/c_0}$  ( $\alpha, \beta \in \mathbb{C}$ )

• Condition at  $x=0$  satisfied if  $A = \alpha + \beta$ , so need one more piece of information to determine  $\alpha$  and  $\beta$ .

• Impose a radiation condition (RC) that the source at  $x=0$  can only produce outward-travelling waves.

• Since  $p' = \alpha e^{i\omega(x-ct)/c_0} + \beta e^{-i\omega(x+ct)/c_0}$ , the radiation condition requires  $\beta = 0$  for  $x > 0$  and  $\alpha = 0$  for  $x < 0$ .



$$\Rightarrow \alpha = A, \beta = 0 \text{ for } x > 0$$

$$\alpha = 0, \beta = A \text{ for } x < 0$$

$$\Rightarrow p' = \begin{cases} \operatorname{Re}(A e^{i\omega(x-ct)/c_0}) & \text{for } x > 0 \\ \operatorname{Re}(A e^{-i\omega(x+ct)/c_0}) & \text{for } x < 0 \end{cases} \quad (+)$$

• NB: If we'd used instead  $p' = F(x-ct) + G(x+ct)$ , then the radiation condition would require

$$p' = \begin{cases} F(x-ct) & \text{for } x \geq 0, \\ G(x+ct) & \text{for } x \leq 0; \end{cases}$$

then the condition at  $x=0$  gives  $F(-ct) = A e^{-i\omega t}$ ,  $G(ct) = A e^{-i\omega t}$  and hence (+).

## Example: 3D spherically symmetric waves

p.9

$$\bullet p' = p'(r, t), \quad r = (x^2 + y^2 + z^2)^{1/2} \Rightarrow p'_{tt} = c_0^2 \left( p'_{rr} + \frac{2}{r} p'_r \right) = \frac{c_0^2}{r} \frac{\partial^2}{\partial r^2} (r p')$$

$$\Rightarrow \frac{\partial^2}{\partial t^2} (r p') = c_0^2 \frac{\partial^2}{\partial r^2} (r p')$$

$$\Rightarrow p' = \frac{F(r - c_0 t) + G(r + c_0 t)}{r}$$

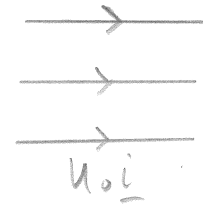
• If waves are generated by a point transducer at origin, then

$$\lim_{r \rightarrow 0^+} r p'(r, t) = A e^{-i\omega t} \quad (\text{real part understood})$$

• Now radiation condition  $\Rightarrow$  only outgoing waves, i.e.  $G = 0$

$$\Rightarrow F(-c_0 t) = A e^{-i\omega t} \Rightarrow p' = \frac{A}{r} e^{i\omega(r - c_0 t)/c_0}$$

## Example: acoustic waves in a background flow



P.10

- Consider barotropic flow governed by ①-③ but with base state given by  $\underline{u} = U_0 \underline{i}$ ,  $\rho = \rho_0$ ,  $p = p_0$ , where  $U_0 \in \mathbb{R}^+$
- Introduce small perturbation:  $\underline{u} = U_0 \underline{i} + \underline{u}'$ ,  $\rho = \rho_0 + \rho'$ ,  $p = p_0 + p'$ .
- Before  $\frac{D}{Dt} = \frac{\partial}{\partial t} + \underline{u}' \cdot \nabla$  replaced by  $\frac{\partial}{\partial t}$  upon linearization.
- Now  $\frac{D}{Dt} = \frac{\partial}{\partial t} + (U_0 \underline{i} + \underline{u}') \cdot \nabla$  replaced by  $\frac{\partial}{\partial t} + U_0 \frac{\partial}{\partial x}$  in ①L - ③L, so assuming irrotational  $\underline{u}' = \nabla \phi$  with  $(\frac{\partial}{\partial t} + U_0 \frac{\partial}{\partial x})^2 \phi = c_0^2 \nabla^2 \phi$ .

Example:  $\phi = \phi(x, t)$

p. 11

$$\bullet \phi = \phi(x, t) \Rightarrow \underbrace{1}_{A} \cdot \phi_{tt} + \underbrace{2U_0}_{2B} \phi_{xt} + \underbrace{(U_0^2 - c_0^2)}_C \phi_{xx} = 0$$

$$\bullet \text{Characteristics: } \frac{dx}{dt} = \lambda, \quad A\lambda^2 - 2B\lambda + C = 0 \Rightarrow \lambda = U_0 \pm c_0$$

$$\bullet \text{Let } \xi = x - (U_0 + c_0)t, \quad \eta = x - (U_0 - c_0)t, \quad \text{then } \phi_{\xi\eta} = 0 \Rightarrow \phi = F(\xi) + G(\eta)$$

$$\bullet \text{Hence, } \phi = \underbrace{F(x - (U_0 + c_0)t)}_{\text{right travelling}} + \underbrace{G(x - (U_0 - c_0)t)}_{\substack{\text{right travelling } U_0 > c_0 \text{ (supersonic)} \\ \text{left travelling } U_0 < c_0 \text{ (subsonic)}}}$$

• Transducer at  $x = 0$  has "zone of silence" upstream ( $x < 0$ ) in supersonic case!

## Example: $\phi = \phi(x, y)$

p.12

- $\phi = \phi(x, y) \Rightarrow (1 - M^2)\phi_{xx} + \phi_{yy} = 0$ ,  $M = \frac{u_0}{c_0}$  is Mach number
- $M < 1$  (subsonic)  $\Rightarrow$  elliptic PDE  $\Rightarrow$  localized disturbances felt everywhere.
- $M > 1$  (supersonic)  $\Rightarrow$  hyperbolic PDE  $\Rightarrow$  localized disturbances propagate at finite speed along characteristics s.t.  $\frac{dy}{dx} = \pm (M^2 - 1)^{-1/2}$ , the general solution being  $\phi = F(y - (M^2 - 1)^{-1/2}x) + G(y + (M^2 - 1)^{-1/2}x)$ .
- Explore later on these dramatic differences & their implications for the theory of flight.