

# Waves and Compressible Flow

## Lecture 11

## 4. Nonlinear waves

- One-dimensional gas dynamics
- Shallow water theory
- Use method of characteristics to construct solutions.

## One-dimensional gas dynamics

- If body forces are negligible, the one-dimensional Euler equations are

$$\rho_t + u\rho_x + \rho u_x = 0, \quad u_t + uu_x = -\frac{1}{\rho} P_x.$$

- Assuming homentropic flow,  $p = k\rho^\gamma$  for constants  $k > 0$  and  $\gamma > 1$ .

- Define speed of sound  $c$  via  $c^2 = \frac{dp}{d\rho} = \gamma k \rho^{\gamma-1} = \frac{\gamma p}{\rho}$ , then

$$\rho = \left(\frac{c^2}{\gamma k}\right)^{\frac{1}{\gamma-1}} = A c^{\frac{2}{\gamma-1}} \text{ and } p = \frac{\rho c^2}{\gamma} = \frac{A}{\gamma} c^{\frac{2\gamma}{\gamma-1}},$$

where  $A = \left(\frac{1}{\gamma k}\right)^{\frac{1}{\gamma-1}}$ , so can eliminate  $\rho$  and  $p$  in favour of  $c$ .

• If  $p = Ac^{\frac{2}{\delta-1}}$ ,  $\rho = \frac{A}{r} c^{\frac{2r}{\delta-1}}$ , then

$$p_t = \frac{2A}{\delta-1} c^{\frac{2}{\delta-1}-1} c_t, \quad p_x = \frac{2A}{\delta-1} c^{\frac{2}{\delta-1}-1} c_x, \quad p_x = \frac{2A}{\delta-1} c^{\frac{2r}{\delta-1}-1} c_x = \frac{2\rho}{\delta-1} c_x$$

$$\bullet \quad p_t + up_x + pu_x = 0 \Rightarrow \frac{2}{\delta-1} (u + u(c_x)) + cu_x = 0 \quad (1)$$

$$u_t + uu_x + \frac{1}{\rho} p_x = 0 \Rightarrow u_t + uu_x + \frac{2}{\delta-1} c c_x = 0 \quad (2)$$

$$\bullet \quad (1) \pm (2) \Rightarrow \frac{\partial}{\partial t} \left( u \pm \frac{2c}{\delta-1} \right) + (u \pm c) \frac{\partial}{\partial x} \left( u \pm \frac{2c}{\delta-1} \right) = 0$$

$$\Rightarrow \left( \frac{\partial}{\partial t} + (u \pm c) \frac{\partial}{\partial x} \right) \left( u \pm \frac{2c}{\delta-1} \right) = 0$$

$$\Rightarrow \frac{d}{dt} \left( u \pm \frac{2c}{\delta-1} \right) = 0 \text{ on curves satisfying } \frac{dx}{dt} = u \pm c$$

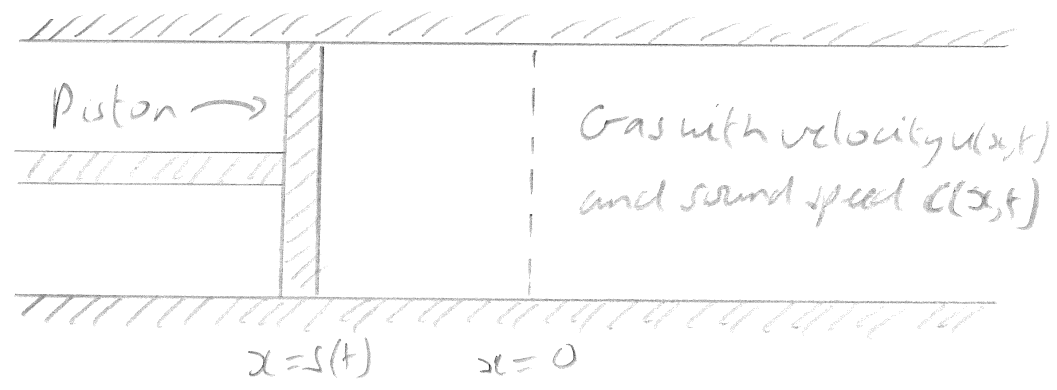
$$\Rightarrow \underline{R_{\pm} = u \pm \frac{2c}{\delta-1}} \text{ are constant on curves on which } \underline{\frac{dx}{dt} = u \pm c}$$

Riemann  
invariants

$\pm$  characteristics

Example: flow due to a piston

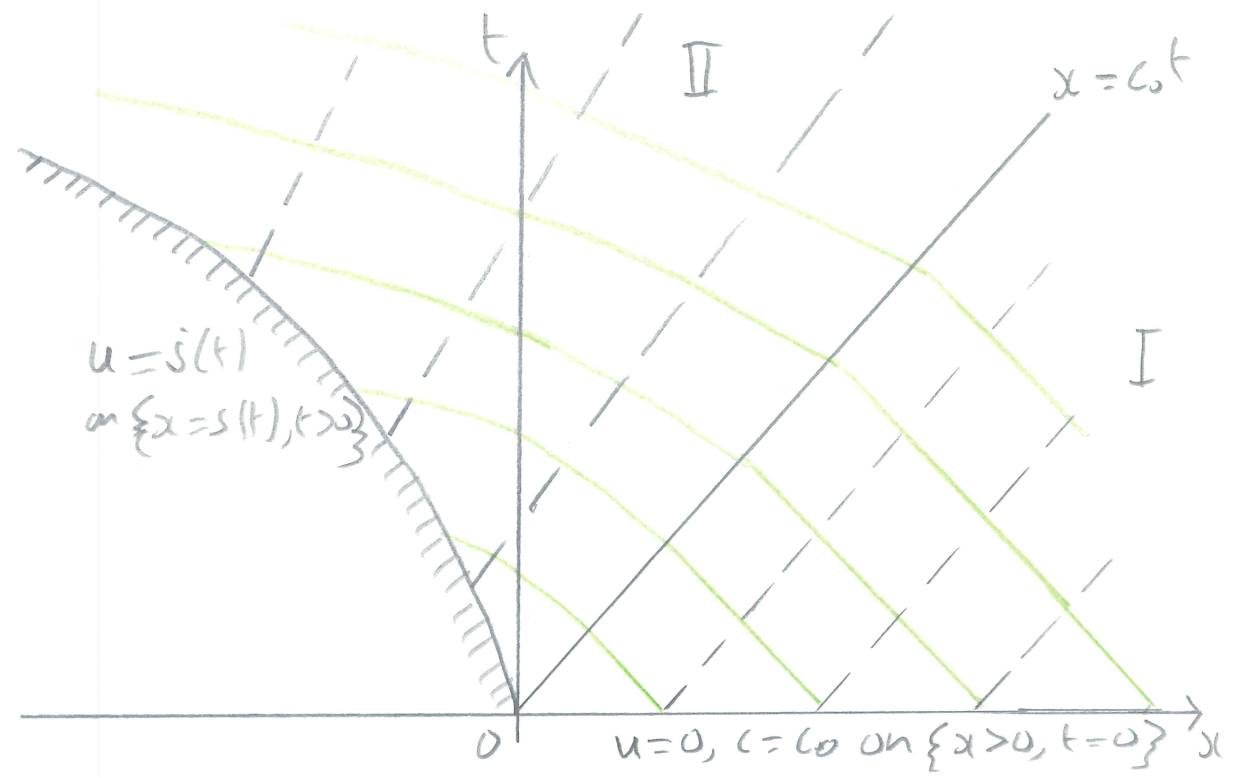
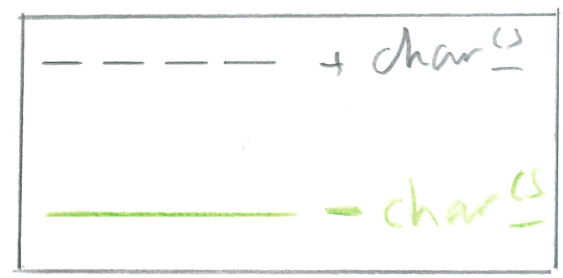
- Suppose the gas is initially at rest in  $x > 0$  with  $p = p_0, \rho = \rho_0, c = c_0 = \left(\frac{\gamma p_0}{\rho_0}\right)^{1/2}$  confined in a tube by a piston initially at  $x = 0$ .
- The piston is then withdrawn so that its position at time  $t$  is  $x = s(t)$ , where  $s(0) = 0$  and  $\dot{s}(t) < 0$  for  $t > 0$ .



• ICs :  $u = 0, c = c_0$  for  $x > 0$  at  $t = 0$ .

• Assuming the piston does not leave the gas behind (i.e. no vacuum forms), the KIR on the piston says  $u = \dot{s}(t)$  on  $x = s(t)$  for  $t > 0$ .

• Characteristic diagram:



Region I

- Consider  $\pm$  characteristics originating from  $\{x > 0, t = 0\}$ .

- Where these characteristics intersect,

$$u \pm \frac{2c}{\gamma-1} = 0 \pm \frac{2c_0}{\gamma-1} \Rightarrow u = 0, c = c_0$$

- Hence, such  $\pm$  characteristics have  $\frac{dx}{dt} = 0 \pm c_0$  and are therefore straight lines.

- They therefore map out  $x > c_0 t, t > 0$ , i.e.  $u = 0, c = c_0$  in region I.

## Region II

- On a + characteristic originating from  $(x, t) = (s(\xi), \xi)$  on the piston,

$$u + \frac{2c}{\gamma-1} = R_+(\xi) = \text{constant.} \quad (+)$$

- Where this + characteristic intersects the family of - characteristics from  $\{x > 0, t = 0\}$ , we also have

$$u - \frac{2c}{\gamma-1} = 0 - \frac{2c_0}{\gamma-1}. \quad (#)$$

- By (+) and (#),  $u$  and  $c$  are constant on this + characteristic, so by the KBC on the piston,  $u = \dot{s}(\xi)$ , and so by (#),  $c = c_0 + \frac{\gamma-1}{2} \dot{s}(\xi)$ .



- Hence, the + characteristic originating from  $(s(\bar{t}), \bar{t})$  on the piston is straight with  $\frac{dx}{dt} = u+c = c_0 + \frac{\delta+1}{2} \dot{s}(\bar{t})$ .
- Since it passes through  $x=s(\bar{t})$  at time  $t=\bar{t}$ , it has equation

$$x - s(\bar{t}) = \left( c_0 + \frac{\delta+1}{2} \dot{s}(\bar{t}) \right) (t - \bar{t}).$$

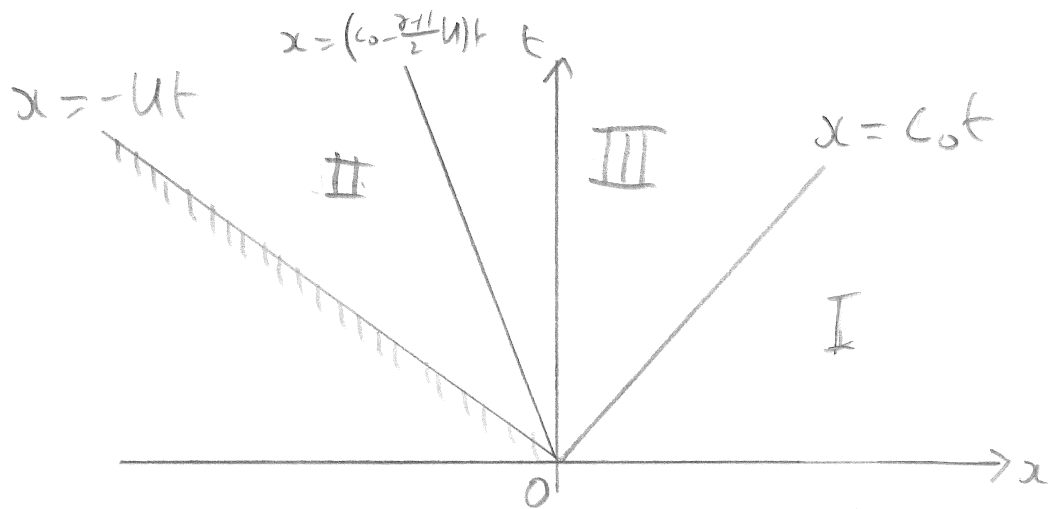
- This gives the parametric solution  $u = \dot{s}(\bar{t})$ ,  $c = c_0 + \frac{\delta-1}{2} \dot{s}(\bar{t})$  and  $x - s(\bar{t}) = \left( c_0 + \frac{\delta+1}{2} \dot{s}(\bar{t}) \right) (t - \bar{t})$  in  $s(t) < x < c_0 t$ ,  $t > 0$ , i.e. in region II, provided  $c \geq 0$  (so that  $p, \rho \geq 0$ ) and the  $\pm$  characteristics fill this region, as illustrated in characteristic diagram above.

- But  $c \geq 0 \Leftrightarrow c_0 + \frac{\delta-1}{2} \dot{s}(z) \geq 0 \Leftrightarrow -\dot{s}(z) \leq \frac{2c_0}{\delta-1}$
- If the withdrawal speed is ever faster than  $\frac{2c_0}{\delta-1}$ , then the piston leaves the gas behind.
- A vacuum then forms near the piston in which  $c = p = 0$ , and it is as if the gas expands freely into a vacuum.
- For example, if  $s(t) = -\alpha t^2/2$ , where  $\alpha$  is the constant acceleration, then the piston leaves the gas behind at time  $t = \frac{2c_0}{(\delta-1)\alpha}$ .

Example with expansion fan

- If  $s(t)$  particularly simple, can eliminate  $\tau$  in region II, e.g. if  $s(\tau) = -U\tau$  ( $U \in \mathbb{R}^+$ ), then  $u = -U$ ,  $c = c_0 - \frac{\gamma-1}{2}U$  where  $x = -U\tau + (c_0 - \frac{\gamma-1}{2}U)(t-\tau)$  for  $\tau \geq 0$ .

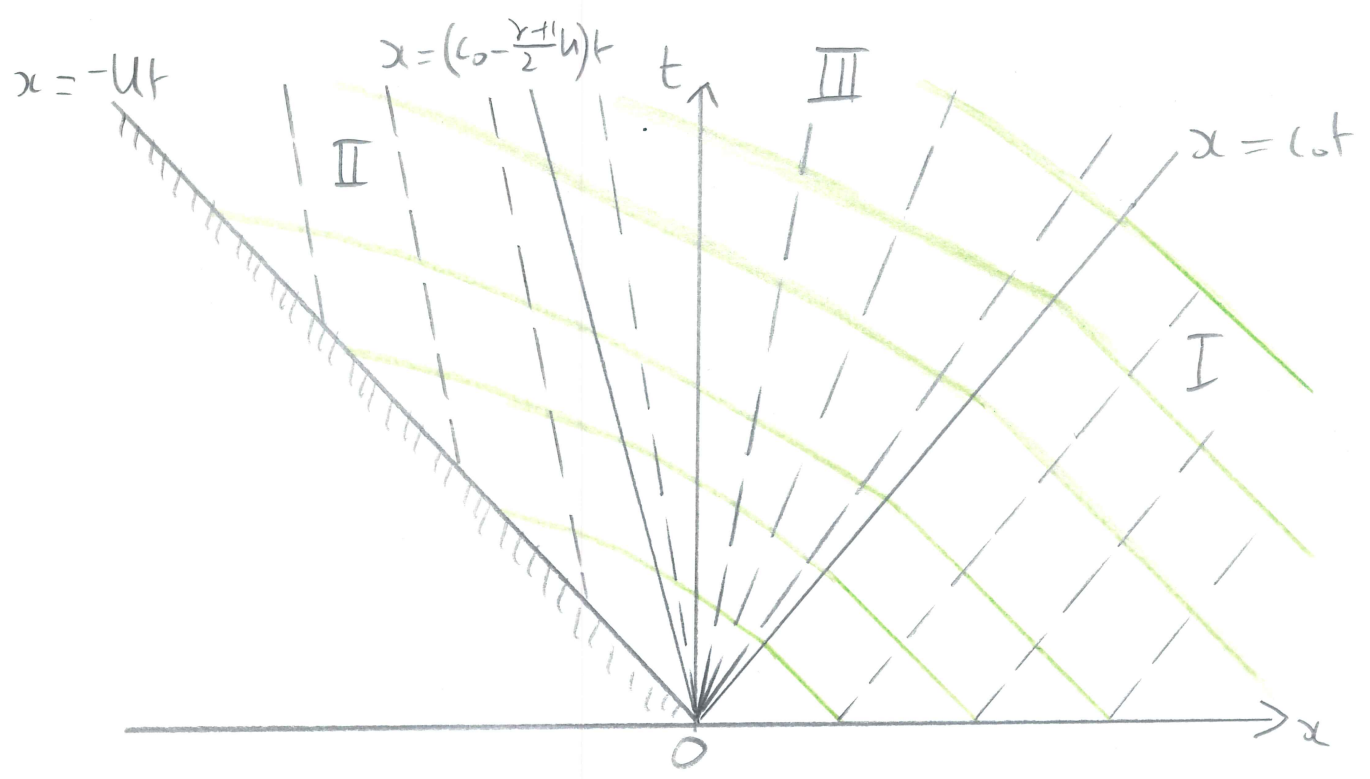
- First such + characteristic has  $\tau = 0 \Rightarrow x = (c_0 - \frac{\gamma-1}{2}U)t < c_0 t!$



What happens in the gap, region II?

- In region III, still have - characteristics from  $\{x > 0, t = 0\}$ , so  $u - \frac{2c}{\delta-1} = \frac{-2c_0}{\delta-1}$  here.
- On a + characteristic  $u + \frac{2c}{\delta-1} = \text{constant}$ , so  $u, c$  constant, so it is straight.
- To avoid it crossing + characteristics in region I and II (which would lead to a contradiction), it must pass through the origin and therefore have **speed**  $\frac{dx}{dt} = u + c = \frac{x}{t}$ .
- Hence,  $u = \frac{2}{\delta+1} \left( \frac{x}{t} - c_0 \right)$ ,  $c = \frac{2c_0 + (\delta-1)x/t}{\delta+1}$  in region III, which occupies  $(c_0 - \frac{\delta+1}{2}u)t < x < -c_0t$ .

- This structure, with the + characteristics radiating out from the origin is known as an expansion fan, as illustrated assuming  $U < \frac{2c_0}{\gamma-1}$ .



- If  $U > \frac{2c_0}{\gamma-1}$ , then there is no region II and the expansion fan stops where  $c = 0$ , i.e. at  $x = -\frac{2c_0 t}{\gamma-1}$  — see online notes.